Monday, 22/10/2007 2:17:15 PM Linda Lacelle

User **Process Sheet** Customer CU-DAR001 Dart Helicopters Services **Drawing Name** : ARM : 35296-13 Job Number : 12882 **Estimate Number** P.O. Number Part Number : D3560042 : 22/10/2007 S.O. No. : This Issue **Drawing Number** - D3560 LIND Prsht Rev. : NC : N/A Project Number First Issue : 11 Type : SMALL /MED FAB : C **Drawing Revision** : 35030 Previous Run Material **Due Date** : 29/10/2007 Written By Qty: Each Checked & Approved By Comment 07.05.24 : Est Rev:A EC est rev B ECN 987 07.10.09 EC verified by: DD Additional Product Job Number: Seq. #: Machine Or Operation: Description: 1.0 M6061T6B0500X05000 6061-T6 Bar .50" x 5.0" Comment: Qty.: 1.4648 f(s)/Unit Total: 20.5065 f(s) 6061-T6 Bar 0.50" x 5.00" V 02/11/30 Batch: M/06/82 2.0 BAND SAW Comment: BAND SAW Cut blanks 16.750" long 3.0 HAAS1 Comment: HAAS CNC VERTICAL MACHINING #1 1- Mill as per Folio FA694 Rev: & Dwg D3560 Rev: \_\_\_\_ 2-C'sink 0.196" hole on manual mill as per dwg D3560 3-Deburr per dwg D3560 4.0 QC2 INSPECT PARTS AS THEY COME OFF MACHINE

PARTS AS THEY COME OFF MACHINE

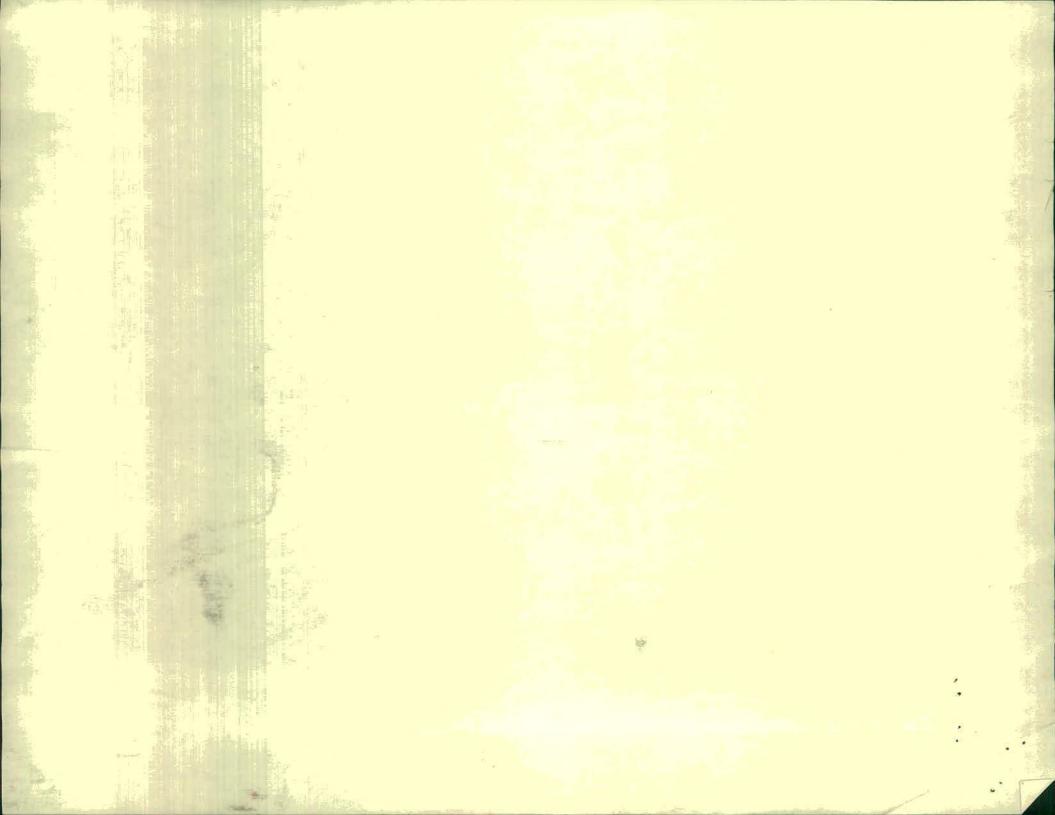
Comment: INSPEC 5.0

QC8 SECOND CHECK



Comment: SECOND CHECK





Monday, 22/10/2007 2:17:15 PM Date: Linda Lacelle User: **Process Sheet** Drawing Name: ARM Customer: CU-DAR001 Dart Helicopters Services Part Number: D3560042 Job Number: 35296 Job Number: Description: Seq. #: Machine Or Operation: PLATE D35921 6.0 Total: 14.0000 Each(s) Comment: Qty.: 1.0000 Each(s)/Unit PLATE LARGE FAB 1 7.0 Comment: LARGE FABRICATION RESOURCE 1 1-Weld assembly as per dwg D3560 STEP: 1- clean material (buff bracket and bottom of arm with blue pad )  $\Longrightarrow$ 2- set up bracket and arm on jig SP 3- preheat bracket and arm with torch SP 4- clean before welding with brush SP 5- set up machine to 135 amps SP 6- weld across bottom and top ends SP 7- reheat with torch ( 157 8- on one side weld from bottom to top half way SP 9- same for other side (half way) SP 10- from half way point weld the rest of the first side (ease off pedal near end) Se 11- same for remaining side (ease off pedal near end) INSPECT WORK TO CURRENT STEP 8.0 QC5 RK TO CURRENT STEP VISUAL WELDING INSPECTION 9.0 Comment: VISUAL WELDING INSPECTION 10.0 HAND FINISHING RESOURCE #1 Comment: HAND FINISHING RESOURCE #1 08.09-63 Chemical Conversion Coat as per QSI 005 4.1



Monday, 22/10/2007 2:17:15 PM Date: **Process Sheet** Linda Lacelle User: Drawing Name: ARM Customer: CU-DAR001 Dart Helicopters Services Part Number: D3560042 Job Number: 35296 Job Number: Description: Machine Or Operation: Seq. #: INSPECT POWDER COAT/CHEMICAL CONVERSION QC3 11.0 Comment: INSPECT POWDER COAT/CHEMICAL CONVERSION Spacer D2808 12.0 14.0000 Each(s) Total: Comment: Qty.: Spacer SMALL FAB 1 13.0 Comment: SMALL & MEDIUM FAB RESOURCE 1 1-Press bushing in D3560 arm per dwg D3562 INSPECT WORK TO CURRENT STEP QC5 14.0 WORK TO CURRENT STEP Comment: INSPECT PACKAGING RESOURCE #1 PACKAGING 1 15.0 Comment: PACKAGING RESOURCE #1 Identify and Stock Location: QC21 16.0 Comment: FINAL INSPECTION/W/O RELEASE U 08.09.18 Job Completion



## Dart Aerospace Ltd

	WORK ORDER CHANGES								
STEP	PROCEDURE CHANGE	Ву	Date	Qty	Approval Chief Eng / Prod Mgr	Approval QC Inspector			
		-			<u> </u>	7.			
				1					
	STEP					Approval			

Part No: D3560-042 PAR #: NA Fault Category: Rod AB. NS NCR: Ves No DQA: Date: 08/09/19

Resolution: CAR - 08-026 Disposition: QA: N/C Closed: Date: 08/09/19

NCR: 35	5296	WORK ORDER NON-CONFORMANCE (NCR)						
		Description of NC	Corrective Action Section B			Verification	Approval	Approval
DATE STEP		Section A	Initial Action Description Chief Eng Chief Eng		Sign & Section C		Approval Chief Eng	Approval QC Inspector
ह व  10	30 Por # 9.0	Derry inspection act of D2808 Bushing Press fit it was found that a b croth in the D3592-1 from welding. RC. too much lest dow was miser at acq welding unspected : Grain was second the weld.	08 01 1>	* See NCR #  SCRAP and Dastony.  AD Replue  Asper emuil From Daws Stephens  and Alan Stocker on 08/6/100  @ 136 pm	SAD	Sorlor læ	abglis oblatio	
	Ž,	The string for steel,		See CARO8-026	J08-05-11			००००००

NOTE: Date & initial all entries

caloatra 467 (80000 C81:1/17

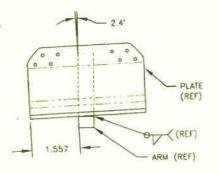
D3560-1 ARM (-041, SHOWN)
OR D3560-2 ARM (-042, OPPOSITE)

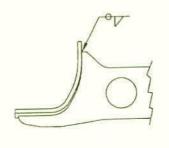
D3592-1 PLATE,
SEE DETAIL A

D3560-041 ARM WELDMENT (SHOWN), D3560-042 ARM WELDMENT (OPPOSITE) 03560-3 ARM (-043, SHOWN)
OR D3560-4 ARM (-044, OPPOSITE)

D3592-1 PLATE,
SEE DETAIL A

D3560-043 ARM WELDMENT (SHOWN).
D3560-044 ARM WELDMENT (OPPOSITE)





DETAIL A (SCALE 1:1) GENERAL NOTES

1) WELD PER QSI 004

- 2) FINISH: CHEMICAL CONVERSION COAT PER DART QSI 005 4.1
- 3) TOLERANCES ARE PER DART QSI 018 UNLESS OTHERWISE NOTED

4) ALL DIMENSIONS ARE IN INCHES



C	07.06.19	REMOVE POWDER COAT
В		REDESIGN AS WELDMENT, ADD POCKETS
A	06.09.25	NEW ISSUE
DESIGN	O DRAWN BY	DART DART AEROSPACE LTD.

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A 06.09.25 NEW ISSUE

DESIGN OF ORANN BY O DART DART AEROSPACE LTD.

HAMMETER AFF. ON TARRO, CAMADA

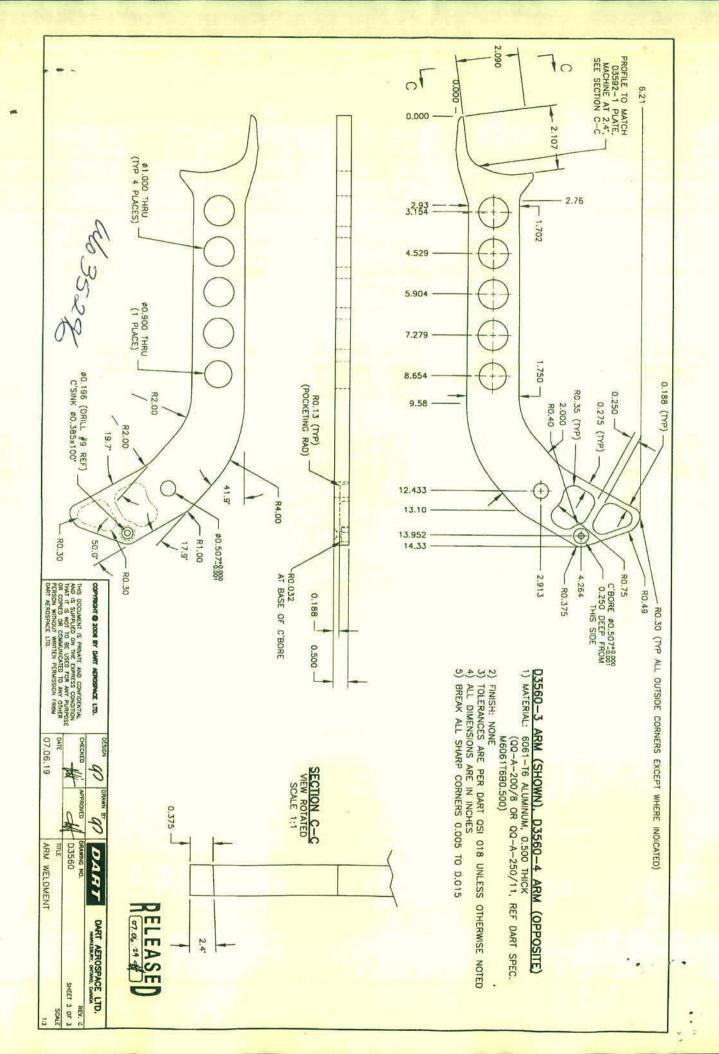
APPROVED DRAWING NO.

DASSO SHEET 1 OF 3

OATE 171LE SCALE

07.06.19

ARM WELDMENT 1.4





From: David Shepherd [mailto:dshepherd@dartaero.com]

Sent: September 10, 2008 1:36 PM

To: 'Alan Stocker'

Cc: 'Chris Provencal'; 'Mike Petsche'; 'Bill Beckett'; 'Susanne Sheldon'

**Subject:** RE: D3560-044 & -042 Cracking

Alan.

Thanks for the pictures.

I am not comfortable with any sort of repair to these parts.

I think that all 14 parts should be scrapped.

And, at the risk of stating the obvious, we need to revisit the manufacturing process of this joint.

My preference, as it was a couple of years ago, is to eliminate this weld.

However, the geometry in that area is a little tricky. Suggest we generate an NCR or PAR or whatever.

David

From: Alan Stocker [mailto:astocker@dartaero.com]
Sent: Wednesday, September 10, 2008 10:26 AM

To: 'David Shepherd'

Cc: 'Chris Provencal'; 'Mike Petsche' Subject: D3560-044 & -042 Cracking

Good morning,

We have 13x D3560-044 and 1x D3560-042 that have cracks all but 1 in the same location. Attached image D3560-044 Crack 1 shows where 13 of the 14 cracks occurred. D3560-044 Crack 2 shows where the other crack occurred. The cracks shown in D3560-044 Crack 1 vary in depth from roughly 3/32 to ½ inch. I discussed this with Chris and Peter the consensus opinion is the parts are scarp. Further discussion with Chris indicates that changing grain direction to 45 degree on the sheet metal part may lower the scrap rate but not eliminate it. This has been done on a previous deviation with a less scrap. D3560-044 Crack 2 appears to just be an anomaly.

Please disposition all 14 parts.

Regards,

Alan Stocker Mechanical Designer

Dart Aerospace Ltd. 1270 Aberdeen Street Hawkesbury, Ontario CANADA K6A 1K7

Phone: 613 632 5200 x 241 FAX: 1 613 632 5246

astocker@dartaero.com

